Date: User: Customer Job Number P.O. Number This Issue Prsht Rev. First Issue Previous Run Written By Comment **Additional Product** Job Number: Seq. #: 1.0 e 345

Monday, 3/26/2007 3:18:57 PM

Kim Johnston

**Process Sheet** 

: CU-DAR001 Dart Helicopters Services

: 31466 : 10386 **Estimate Number** 

: hm

S.O. No. : NIA : 3/26/2007

: NC

: NM

: 31273

: MACHINED PARTS Type

Checked & Approved By

: Est Rev. A

New Issue 05-11-08

JLM

: BAR **Drawing Name** 

Part Number **Drawing Number** 

Material

**Due Date** 

: D31961 . D3196 REVC

: N/A Project Number : C **Drawing Revision** 

: NIA

: 4/6/2007

Qty:

20 Um:

Each



Machine Or Operation:

Description:

6061-T6 Bar .75" X 1.5"



Comment: Qty.:

2.2922 f(s)/Unit Total: 45.8430 f(s)

Material: 6061-T6/T651 (QQ-A-200/8) or (QQ-A-225/8) M104057X11 parts

(M6061T6B0.750x01.500)

Identify for D3196-1 Batch:

Comment: BAND SAW

BAND SAW

Cut blank: (0.75" x 1.50") x 26.200" long Bar

oxloyed x9pms.

3.0

2.0

HAAS CNC VERTICAL MACHINING #1



Comment: HAAS CNC VERTICAL MACHINING #1

1-Machine D3196-1 as per Folio FA339 and Dwg D3196Identify as D3196-1

2-Deburr

4.0 QC2

INSPECT PARTS AS THEY COME OFF MACHINE



Comment: INSPECT PARTS AS THEY COME OFF MACHINE

5.0

SECOND CHECK



Comment: SECOND CHECK

# **Dart Aerospace Ltd**

W/O:		WORK ORDER	CHANGES ,	رْباً	K P		
DATE	STEP	PROCEDURE CHANGE	Ву	Date	Qty	Approval Chief Eng / Prod Mgr	Approval QC Inspector
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	·	<b>₹</b>		•			
				.,	د		
			, en		6	5	

Part No: <u>D3196</u>	PAR #:	_ Fault Category:	NCR: Ye	es No DQA:	Date: Moyla
			QA	: N/C Closed:	Date:

NCR: 3	1460	W	ORK OR	DER NON-CONFORMANCI	E (NCR)	•		
		Description of NC		Corrective Action Section B		Verification	Approval	Approval
DATE	STEP	Section A	Initial Chief Eng	Action Description Chief Eng	Sign & Date	Section C	Chief Eng	QC Inspector
Q.03.04	3	1.100 DIM IS 1.084	P	PART ACCEPTABLE DUES	(		4	
-210 3.07			67.63.04 pr	NUT AFFECT STRESS	S.	/ /	67.03.04	'/
			QS1 642	REPURT. for programy	02/04/04	orary	Dr. 05/1047	Poses of
<b>ડ</b> ટ.ज.ળ	3	(151NK \$0.610 100"	W	PARTACCEPTABLE	1/0/		4)	
		CISINK \$0.610 PIQUE OFY (Z) AFFECTED	08.67.67	PARTACCEPTABLE REF DS EMAIL	MIP	/	08.67.07 pe	foot
			95142			180AT7	QSIUTZ	1000
						,		

NOTE: Date & initial all entries

Monday, 3/26/2007 3:18:57 PM Date: Kim Johnston User: **Process Sheet** Drawing Name: BAR Customer: CU-DAR001 Dart Helicopters Services Job Number: 31466 Part Number: D31961 Job Number: Description: Seq. #: Machine Or Operation: HAND FINISHING1 HAND FINISHING RESOURCE #1 6.0 Comment: HAND FINISHING RESOURCE #1 Chemical Conversion Coat as per QSI 005 4.1 POWDER COATING 7.0 M10160 Comment: POWDER COATING Powder Coat Grey Sandtex (Ref: 4.3.5.6) as per QSI 005 4.3 INSPECT POWDER COAT/CHEMICAL CONVERSION 8.0 ... Comment: INSPECT POWDER COAT/CHEMICAL CONVERSION 9.0 PACKAGING 1 Comment: PACKAGING RESOURCE #1 Identify and Stock Location: 10.0 QC21 Je Part. 20 Comment: FINAL INSPECTION/W/O RELEASE Job Completion

# **Dart Aerospace Ltd**

W/O:		WORK ORDER CHANGES						
DATE	STEP PROCEDUR		PROCEDURE CHANGE		Date	Qty	Approval Chief Eng / Prod Mgr	Approva QC Inspector
			4					
					- · · · · · · · · · · · · · · · · · · ·			
Part No	_	PAR #:	Fault Category:	NCR: Yes			Date:	

Part No:	PAR #:	Fault Category:	NCR: Yes No	DQA:	Date:	
			QA: N/C C	losed:	Date:	

NCR:		WORK ORDER NON-CONFORMANCE (NCR)						
		Description of NC		Corrective Action Section B				
DATE	STEP	Section A	Initial Chief Eng	Initial Action Description Sign &		Verification Section C	Approval Chief Eng	Approval QC Inspector
	,							
							3	

NOTE: Date & initial all entries

DART AEROSPACE LTD

Description: Bar

Part Number: D3196-1

Inspection Dwg: D3196 Rev: C Page 1 of 1

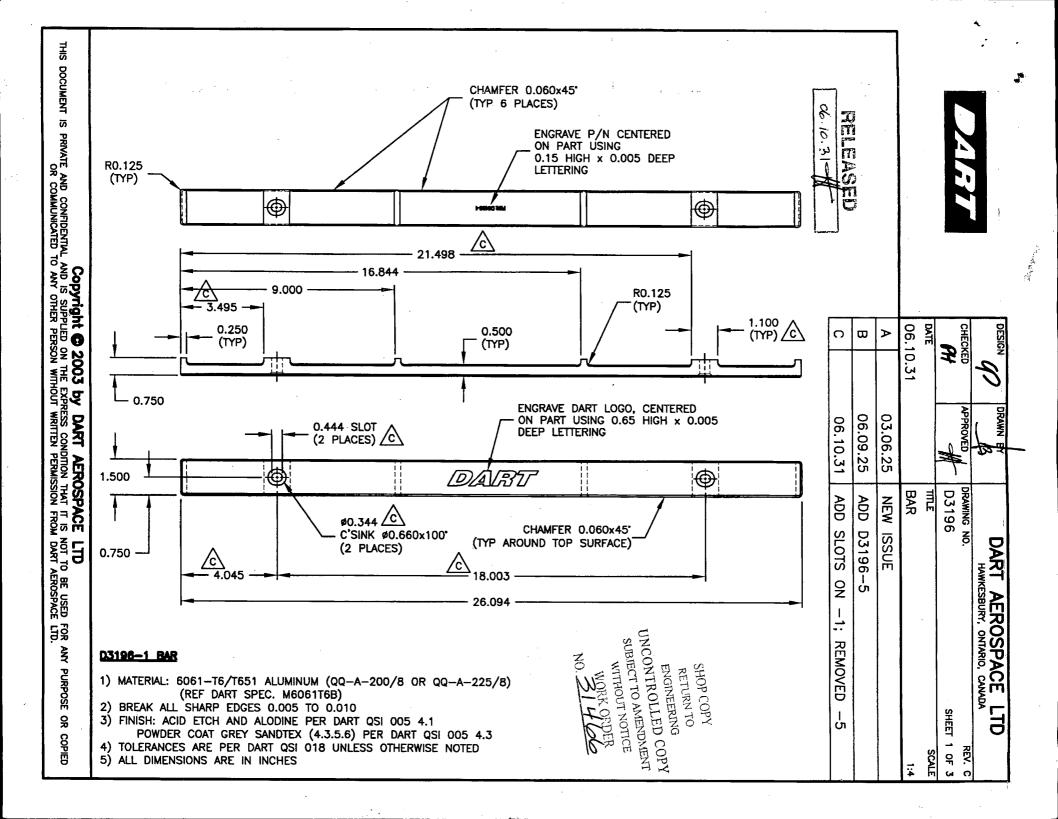
# FIRST ARTICLE INSPECTION CHECKLIST

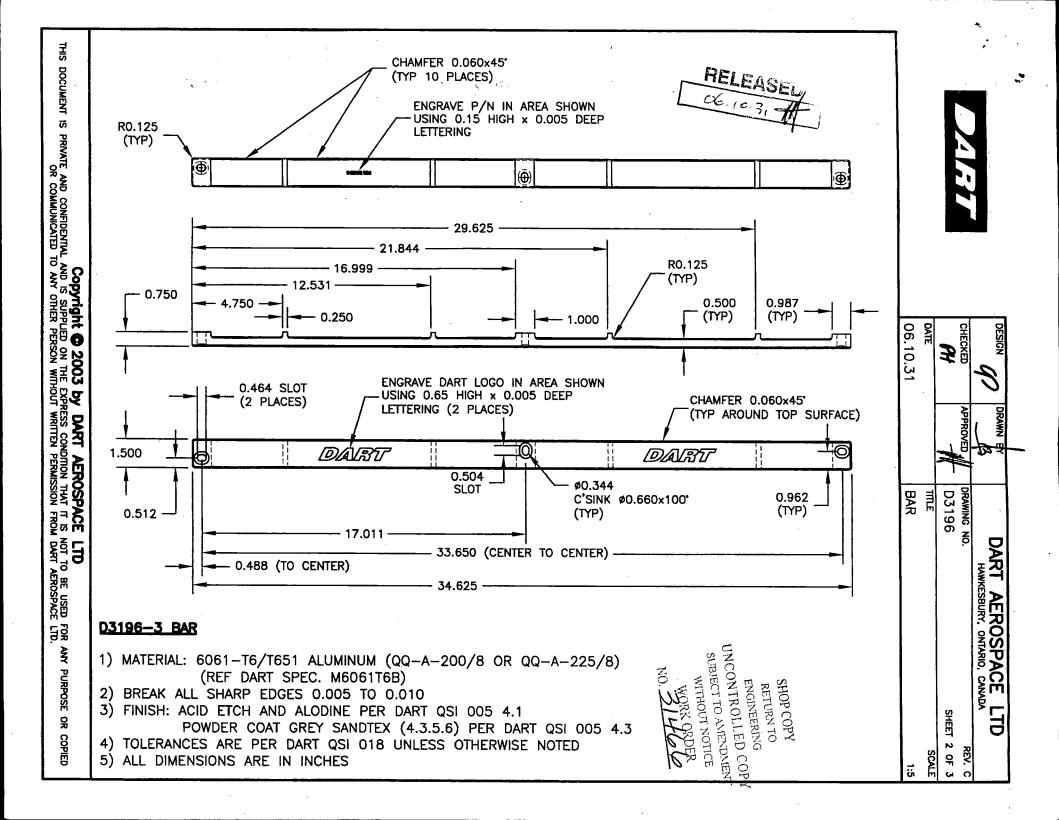
X First Article Prototype

	Drawing Dimension	Tolerance	Actual Dimension	Accept	Reject	Method of Inspection	Comments
1	26.094	+/-0.010	26.094	1	<u>.</u>	TADE MEA	ser:
1	4.045	+/-0.010	4.042				
	18.003	+/-0:005	18,000		. •	TAPE WEASO	se
	, 0.750	1000 +/-0.005 6/304	0.745	/		man saga	tipes
	1.500	+/-0.010	1.492	/		MAL. WI	dth.
/	Ø0.344	+0.006/-0.001	0.344		<del>~ .</del>		
9	Ø0.660 x 100°	+0.008/-0.001 x 0.5°	0.660		i		
	0.060 x 45°	+/-0.010 x 0.5°	0-660 445		-		
	0.750	+/-0.010	6.745		~	MAL. Thek	sec (
1	0.250	+/-0.010	0.245				
1	3.495	+/-0.010	3.494		_	:	
	9.000	+/-0.010	8.998				
	16.844	+/-0.010	16.844				
	- 21.498	+/-0.010	21.448				
	R0.125	+/-0.010	0.175				
	1.1 00	+/-0.010	1.004	See 13 A	h		
	R0.125	+/-0.010	0.125				
	0.444	+/-0.010	0.442				

Measured by:	Audited by: 50	Prototype Approval:	N/A
Date: 076464	Date: 01.04.04	Date:	N/A

Rev	Date	Change	Revised by	Approved
A	04.04.20	New Issue	KJ/RF	
В	06.10.24	Dwg Rev. updated	KJ/JLM 1.A	
С	07.03.21	Dimensions updated per Dwg rev. C	KJ/JLM	B





#### 4.0 Analysis

### 4.1 D3196-1/-3/-4 Bar Analysis

#### 4.1.1 D3196-1 Bar Bending Failure

The loading of the D3196-1 Bar is shown in Figure 1 of Appendix B. The worst case loading is the 16 g forward acting load because the magnitude of the load is higher and the section is smaller in the fwd-aft direction (16g) than it is in the up-down direction (4g).

M	1413 - 0.42	margin or early (eminate)	
$MS := \frac{Mu}{M} - 1$	MS = 0.49	Margin of Safety (Ultimate)	) )
Mu = Fbul ·	Mu = 3057 •in ·1b	Allowable Bending Moment (Ultimate)	
$Mu = Fbul \cdot \frac{2 \cdot I}{2}$	3.5	Modulus of Rupture (Brunn C3.11)  Allowable Reading Memort (Ultimate)	
$Fbu1 = Ftu1 + Fo1 \cdot (k-1)$	Fbu1 = 48911 •psi		)
12		colength of fal	_
$I := \frac{1}{12} \cdot b \cdot t^3$	$I = 0.01563 \cdot in^4$	Inertia of cross section	
M = 2048·in·lb	$M = 2048 \cdot in \cdot lb$	Maximum Ultimate Bending Moment Include 4cbs	
k = 1.5		Shape Factor (Bruhn C3.3) dies nut	
b := 1.50 in t := 0.50 in		Width of Section in Bending  Thickness of Section in Bending  Thickness	

### 4.1.2 D3196-3/-4 Bar Bending Failure

The loading of the D3196-3/-4 Bar is shown in Figure 2 of Appendix B. The worst case loading is the 8g sideways acting load because the magnitude of the load is higher and the section is smaller in the lateral direction (8g) than it is in the up-down direction (4g).

b := 1.50 in t := 0.375 in		Width of Section in Bending Thickness of Section in Bending
k = 1.5		Shape Factor (Bruhn C3.3)
M = 1180·in·lb	$M = 1180 \cdot in \cdot lb$	Maximum Ultimate Bending Moment
$I := \frac{1}{12} \cdot \mathbf{b} \cdot \mathbf{t}^3$	$I = 0.00659 \cdot in^4$	Inertia of cross section
Fbu1 = Ftu1 + Fo1· $(k-1)$	Fbu1 = 48911 •psi	Modulus of Rupture (Bruhn C3.11)
$Mu = Fbu1 \cdot \frac{2 \cdot I}{t}$	Mu = 1720 • in lb	Allowable Bending Moment (Ultimate)
$MS := \frac{Mu}{M} - 1$	MS = 0.46	Margin of Safety (Ultimate)

### **Chris Provencal**

From: David Shepherd [dshepherd@dartaero.com]

**Sent:** May 30, 2008 9:52 AM

To: 'Chris Provencal'
Cc: 'Mike Petsche'

Subject: RE: NCR D3196-1/-3/-4

#### Chris,

I agree with you. The parts are acceptable. I wonder how they had the flexibility to screw around with the CNC program? Guess we better make it crystal clear on the drawing.

David

From: Chris Provencal [mailto:cprovencal@dartaero.com]

Sent: Thursday, May 29, 2008 8:58 AM

**To:** 'David Shepherd' **Cc:** 'Mike Petsche'

Subject: NCR D3196-1/-3/-4

David,

RE: D3196-1/-3/-4,

Checking into an NCR for some -3's they are currently making, I found out there has been confusion as to the direction of the 0.660 dim in the Ø0.660x100° c'sink. They were modifying the program to make the dim the other way, which made the c'sink too small.

I checked stock. We have some D3196-1 in stock, the c'sink on one is 0.664" (good), the rest are about 0.610" (min). The MS spec gives the range of screw head widths as 0.617 - 0.635, which is the 'sharp' dimension, ie the dim to keep the top of the head below the part surface (The edge of the head is chamfered, and the actual max diameter of the screw is about 0.590"). The two listed dims are from the same batch, so I think its more than just confusion over the dwg dim. I made an NCR to ensure the programs are correct, and to add a ref dim to show the direction of the 0.660" (or otherwise clarify it). Chances are, we've shipped a bunch like this.

I attached a bar to the a/c using MS24694-S154 screws, through the fireshield's metal grommet, and it bolted up OK.

I think the deviation is acceptable. Attached pics (one showing good, one bad) show that even though it may stick up slightly above the surface, the actual corners of the screw don't go past the edges of the c'sink. The belt shouldn't fray over this bolt as the bolt holes are located on the solid legs of the bar, and the belts don't go straight down, they'd be coming out to go over the occupant's shoulders. I would have them get the w/o's for these bars and sign of deviations using the 0.610" dim as the minimum allowable.

What do you think?

-Chris